

Starlight suppression technologies from the LUVOIR and HabEx reports

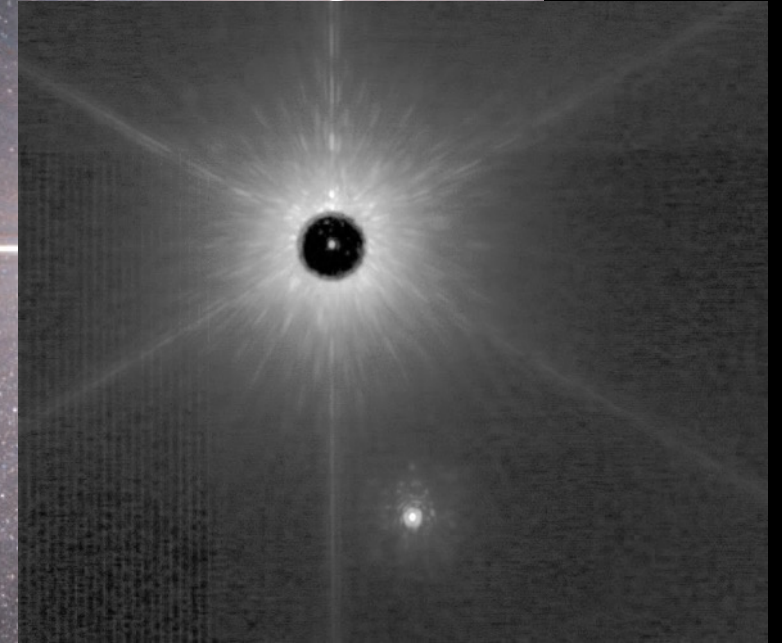
Dr. Rhonda Morgan

Jet Propulsion Laboratory, California Institute of Technology

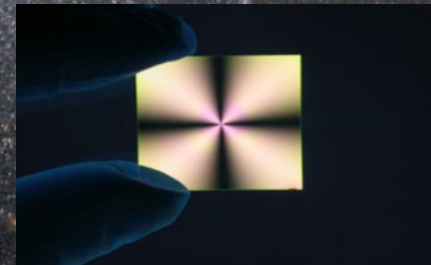
Acknowledgements to the LUVOIR and HabEx Design Teams

*Starlight Suppression Technologies for HWO Flagship
Seattle AAS meeting Splinter Session January 10, 2023*

External Occulters (Starshades)

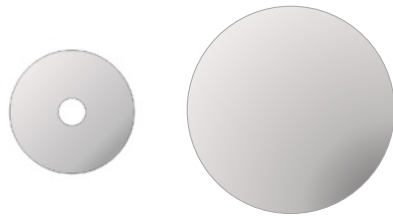


Internal Occulters (Coronagraphs)



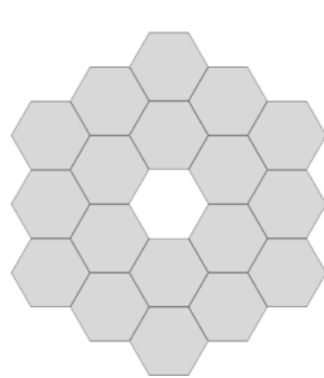
In-space high contrast imaging architectures

	HST	WFIRST	HabEx	JWST	LUVOIR-B	LUVOIR-A
Coronagraph contrast	10^{-5}	10^{-8}	10^{-10}	10^{-6}	10^{-10}	10^{-10}
Starshade diameter		(26 m)	52 m			

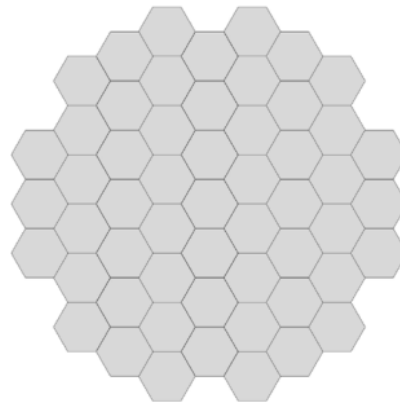


HST/WFIRST
2.4m Primary Mirror
On-Axis Design

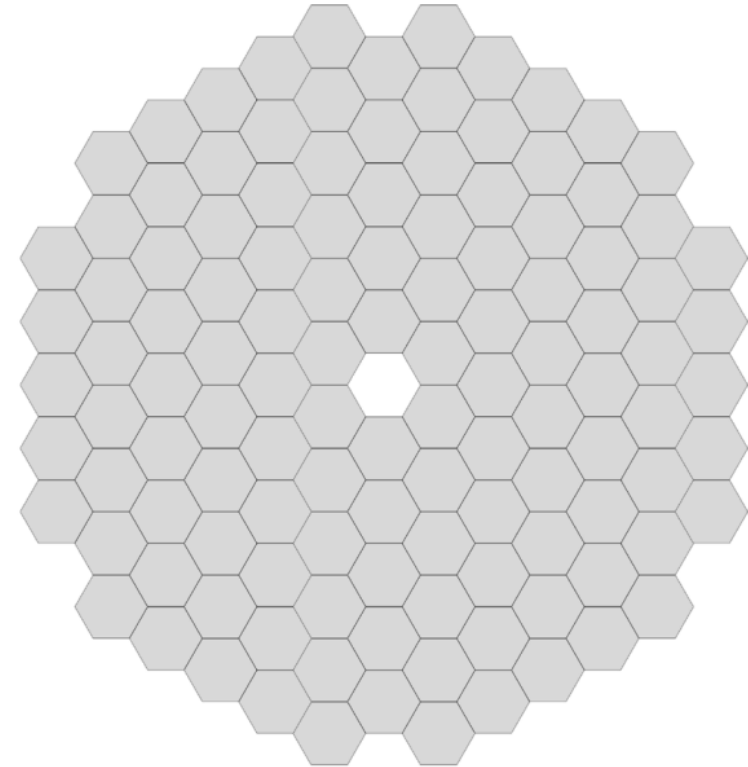
HABEX
4 m Primary Mirror
Off-Axis Design



JWST
6.5m Primary Mirror
On-Axis Design

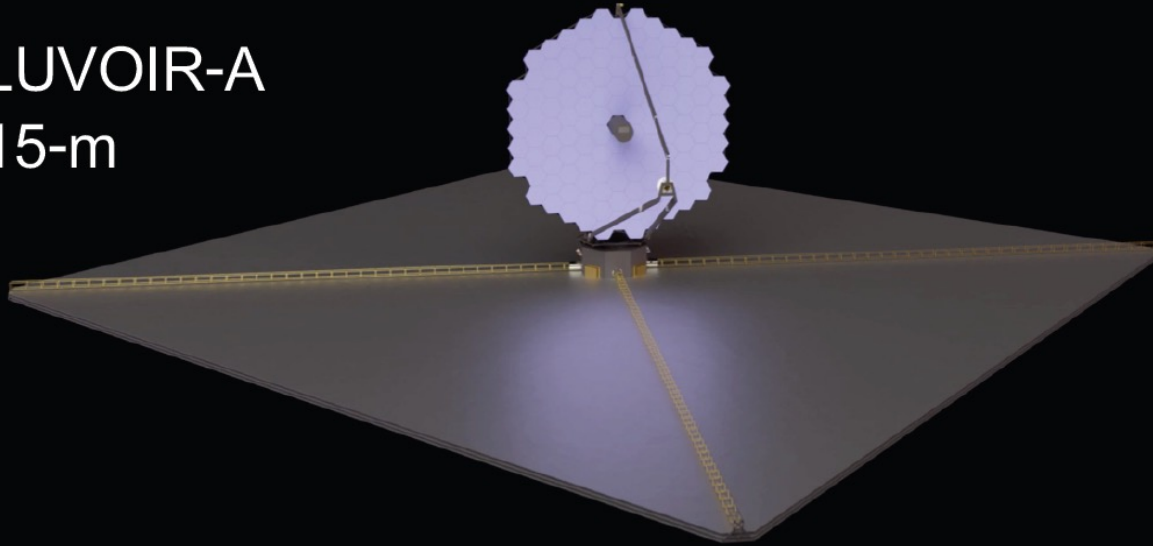


LUVOIR-B
8m Primary Mirror
Off-Axis Design

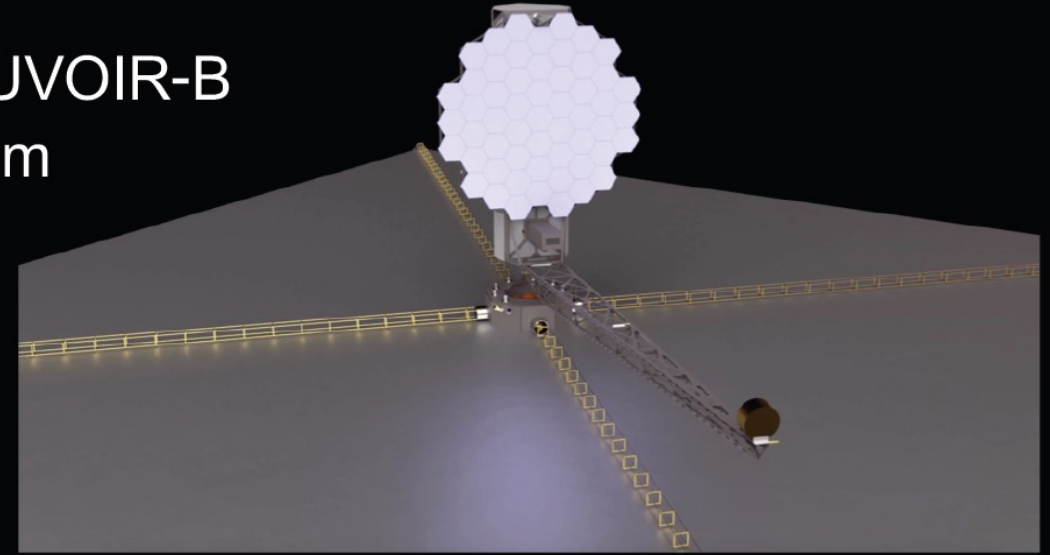


LUVOIR-A
15m Primary Mirror
On-Axis Design

LUVOIR-A
15-m



LUVOIR-B
8-m



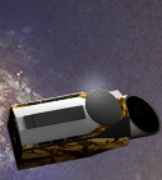
On-axis obscuration is challenging for coronagraph

Coronagraph with 10^{-10} raw contrast

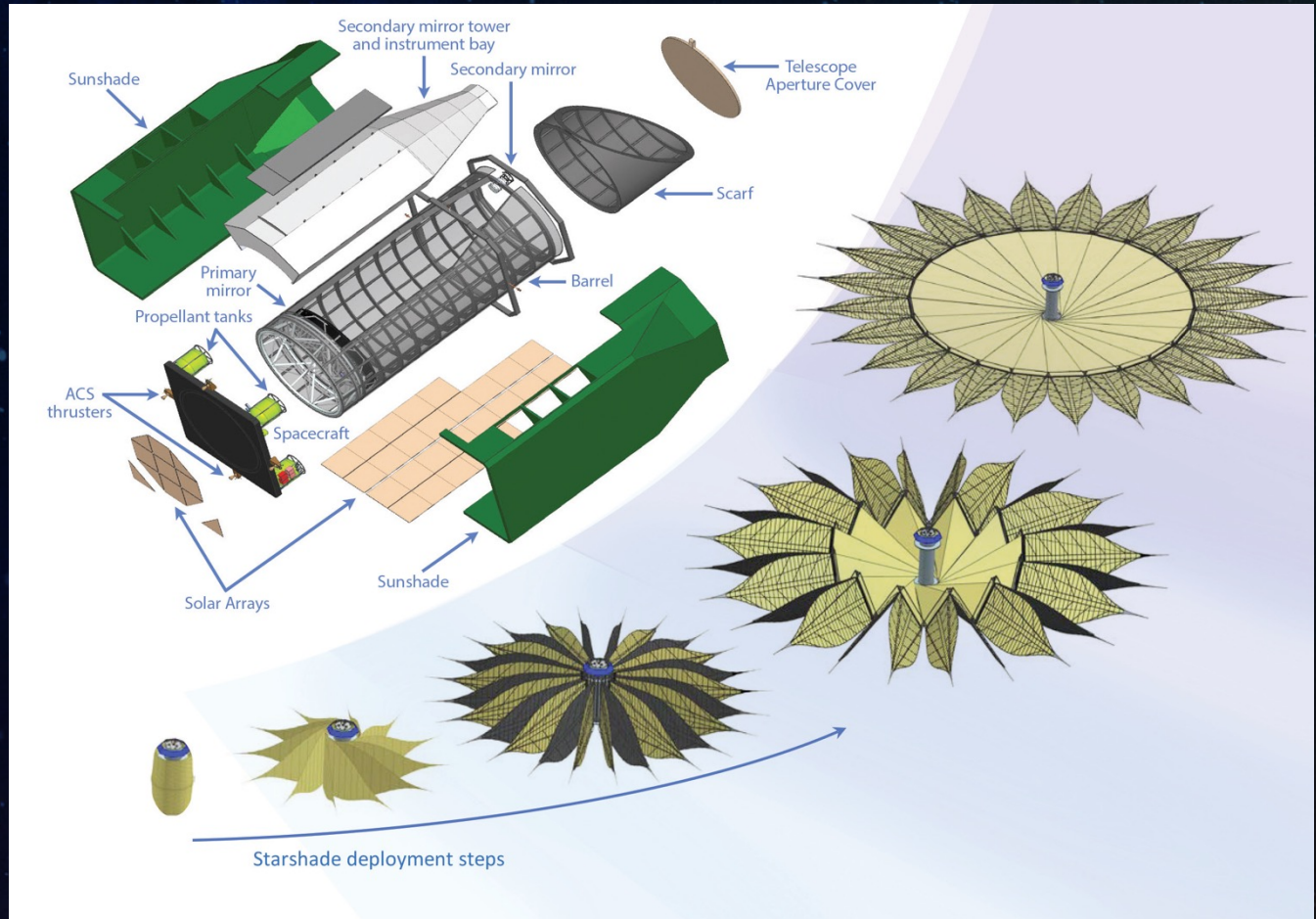
- LUVOIR-A: APLC, LUVOIR-B: VVC
- DM: 128 x 128 MEMS
- Imager: 1024 pixels, IFS: 4096 pixels
- IWA – OWA: 3.5 – 64 I/D
- Wavelengths: 200-525, 515-1030, 1000-2000 nm

Ultra-stable segmented mirror

- Closed back ULE glass segments
- Rigid body actuated segments
- Edge sensors
- Laser metrology truss
- Vibration isolation



Mission Duration	5 years (10 years consumables)
Orbit	Earth-Sun L2 Halo
Telescope Type	Off-axis three-mirror anastigmat
Primary Mirror	4-meter monolith glass-ceramic (Zerodur) substrate with Al+MgF ₂ coating
Attitude Control	Slewing: hydrazine thrusters Pointing: micro-thrusters
Launch Vehicles	Telescope: SLS Block 1-B Starshade: Falcon Heavy
Science Instruments	<ul style="list-style-type: none"> Exoplanet Science <ul style="list-style-type: none"> - Coronagraph - Starshade Observatory Science: <ul style="list-style-type: none"> - UV Spectrograph - Workhorse Camera

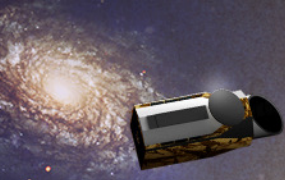


Coronagraph IWA – OWA: 2.4 – 32 I/D, Vis, NIR
 Starshade IWA -OWA: 58 mas – 6" for 300-1000 nm
 IFS detector: 4096 pixels

The complementary strengths of coronagraphs and starshades

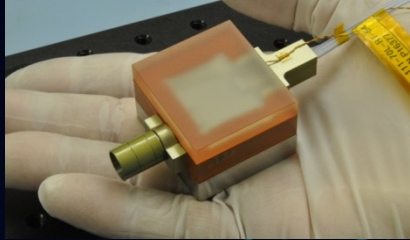
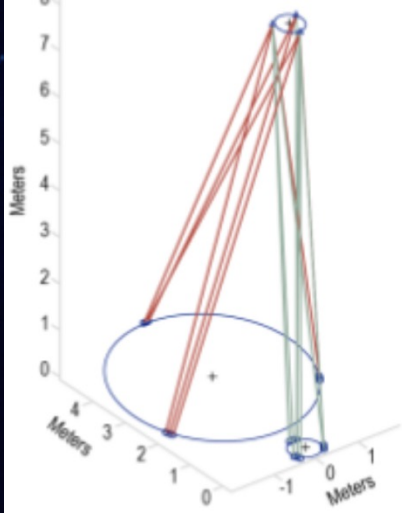
	coronagraph	starshade
Bandwidth	20%	>100%
Throughput	20-40%	~100%
Polarization	Single channel	Unaffected
IWA	2.4 λ/D	1.5 λ/D
OWA	64 λ/D	>235 λ/D
Slew time	<1 hr	~1 week
Starlight suppression	Inside telescope	In front of telescope
Strength	Search survey	High quality spectra

- **Coronagraphs are agile and well-suited for search surveys**
- **Starshades are optimal for high quality spectra due to their smaller IWA, higher throughput, and very broad bandwidth**



Systems level solutions required for coronagraph

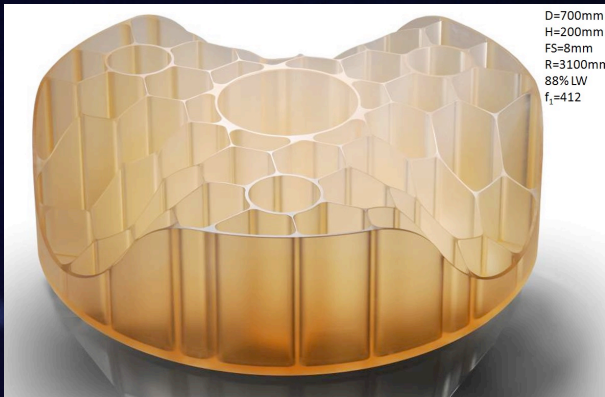
Laser Metrology



TRL 5

Laser Metrology System: Beam launcher, ring laser, and phase meter

TRL 4



D=700mm
H=200mm
FS=8mm
R=3100mm
88% LW
f_r=412

Large Monolith Mirror Fabrication



Microthrusters



ESA/LISA Pathfinder

TRL 4

2.4 m Coating Coating Uniformity



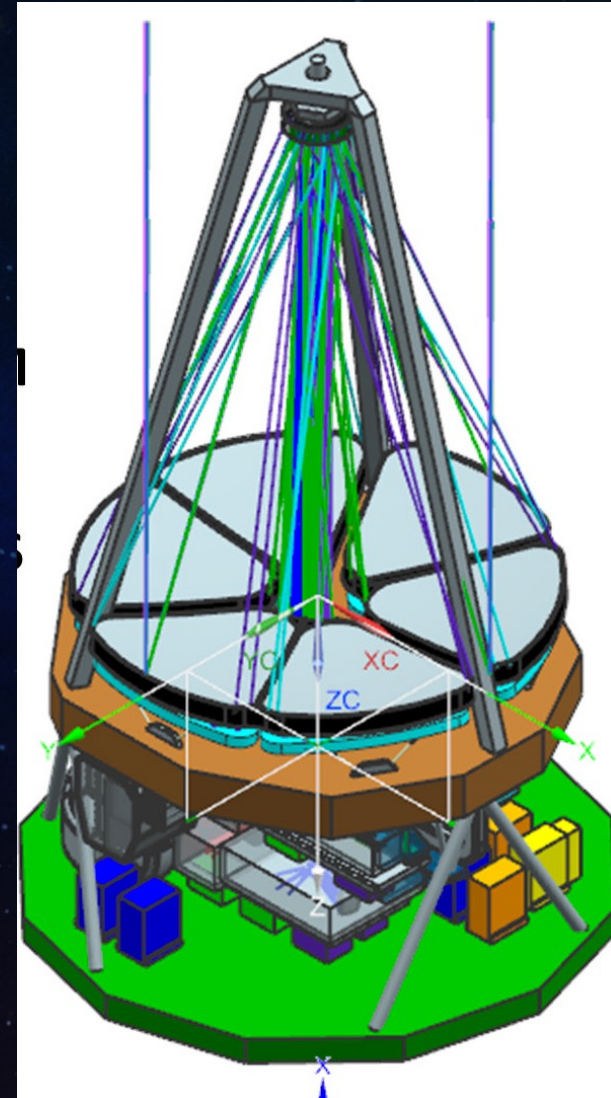


- **No coronagraph**
 - Telescope WFE stability tolerances relaxed 1000 times
 - Starshade provides the highest quality exoplanet spectroscopy
 - But – lower yield of exo-Earths *unless detected before HabEx*

- **Active Optics On-axis Telescope**
 - Corrects Static PM WFE in orbit
 - Segmented to stay within current practice and largest ULE mirrors
 - Laser MET to continuously maintain optical alignment
 - **Lighter (2T) & Smaller Telescope**
 - Light weight ULE (5cm thick) Primary Mirror
 - Total launch Mass = 7.3 T, fits in Delta IV Heavy or Vulcan Centaur
 - More compact (f/1.3)
 - Non deployable OTA a priori scalable to 4m and above

- **Lower cost option**

Estimated Cost Reductions	HabEx 4H	HabEx 3.2S
Smaller Telescope	–	-0.6 \$B
No Coronagraph	–	-0.4 \$B
Smaller Launch Vehicle	–	-0.4 \$B
Same Starshade System	–	–
Lower Reserves	–	-0.4 \$B
Total Estimated Cost	6.8 \$B	5.0 \$B

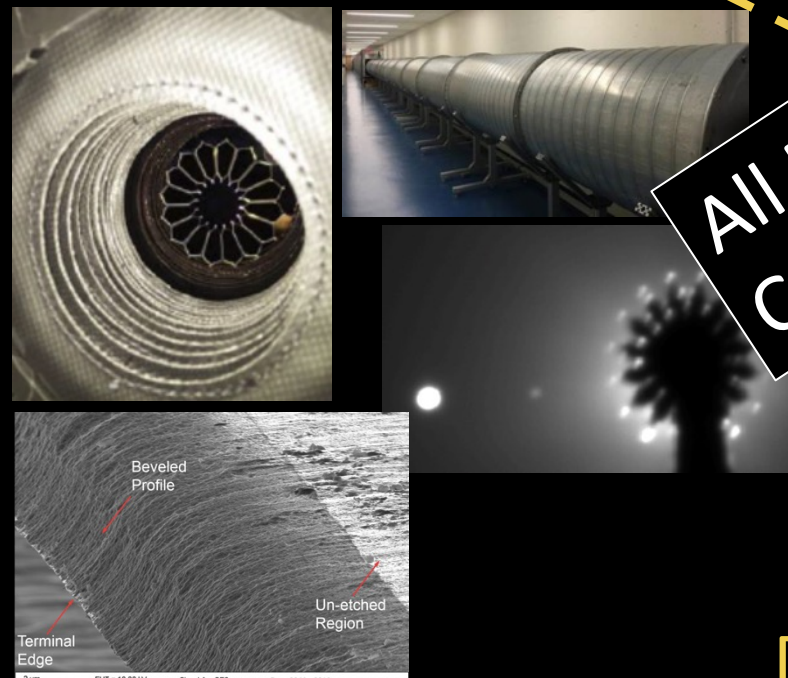


HabEx Report Appendix B, Fig B.1-6



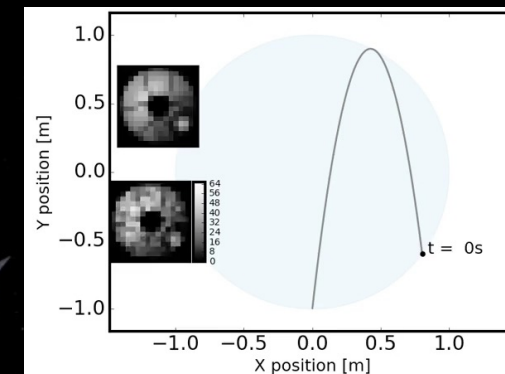
Starshade Technology Gaps

Starlight Suppression



**Is a starshade
required for
UV?**

Formation Sensing and Control



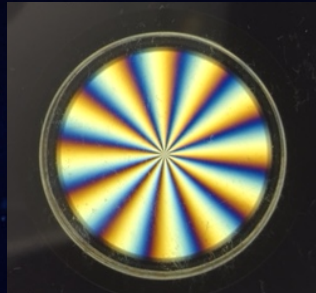
Deployment Accuracy and Shape Stability

Critical Milestones Complete, High-Fidelity Milestones In Progress!

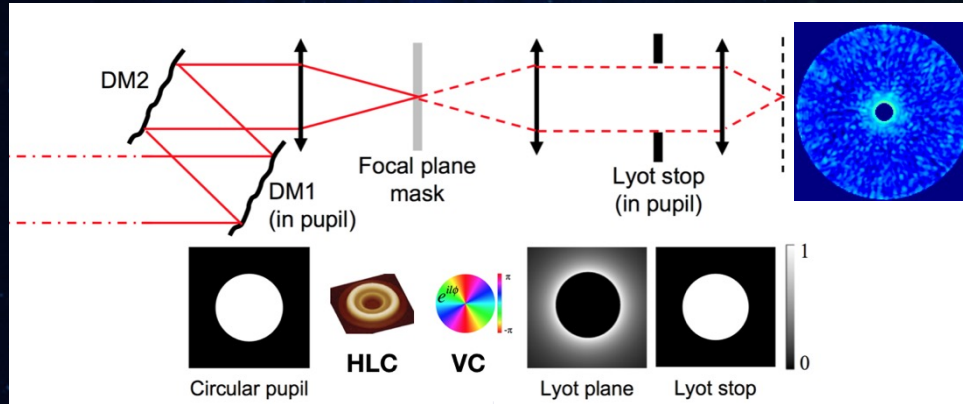




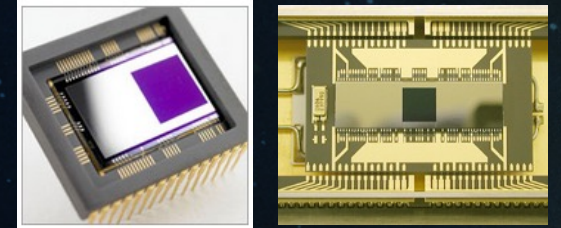
Contrast



TRL 4

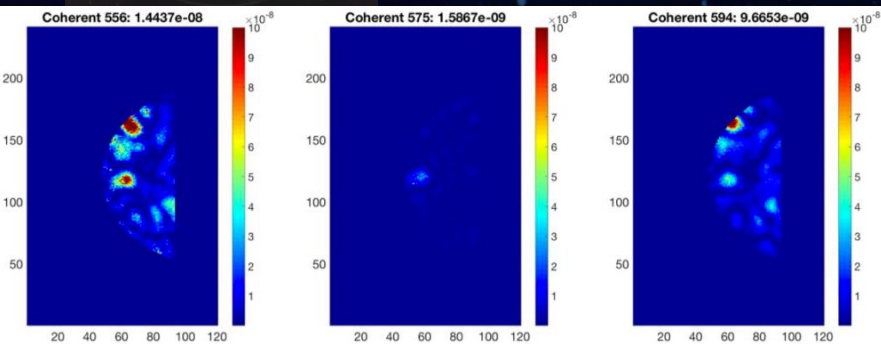


Detection Sensitivity



TRL 4

Ultra-low Noise UV and Visible EMCCD



Coronagraph Architecture

TRL 5



48x48 WFIRST

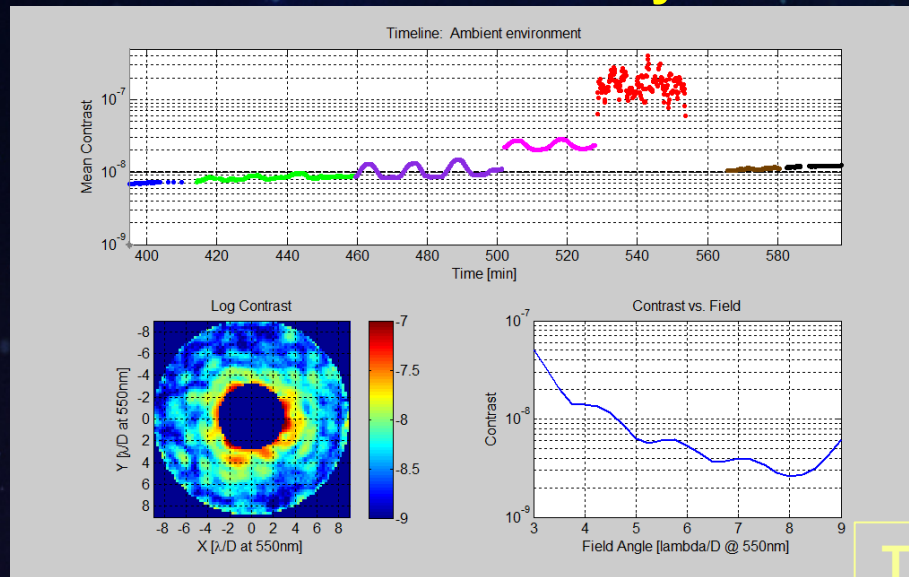
Deformable Mirrors



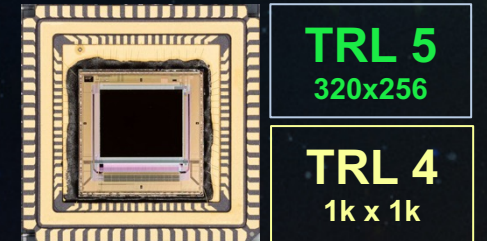
TRL 4

64x64 MEMS

Contrast Stability



Low-Order Wavefront Sensing and Control



TRL 5
320x256

TRL 4
1k x 1k

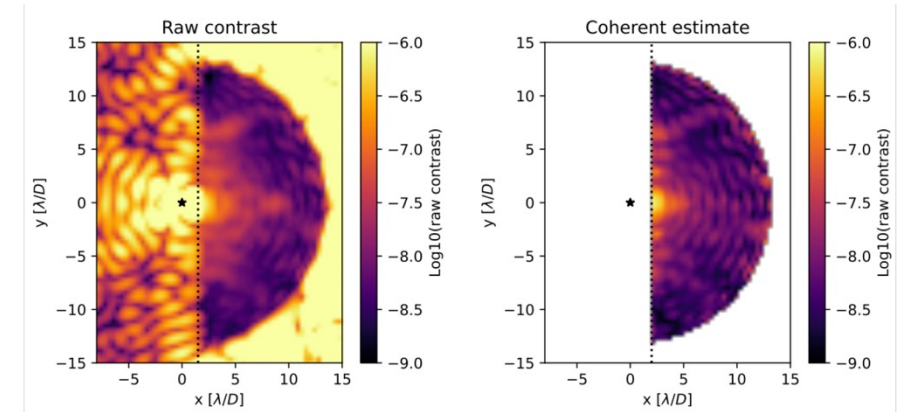
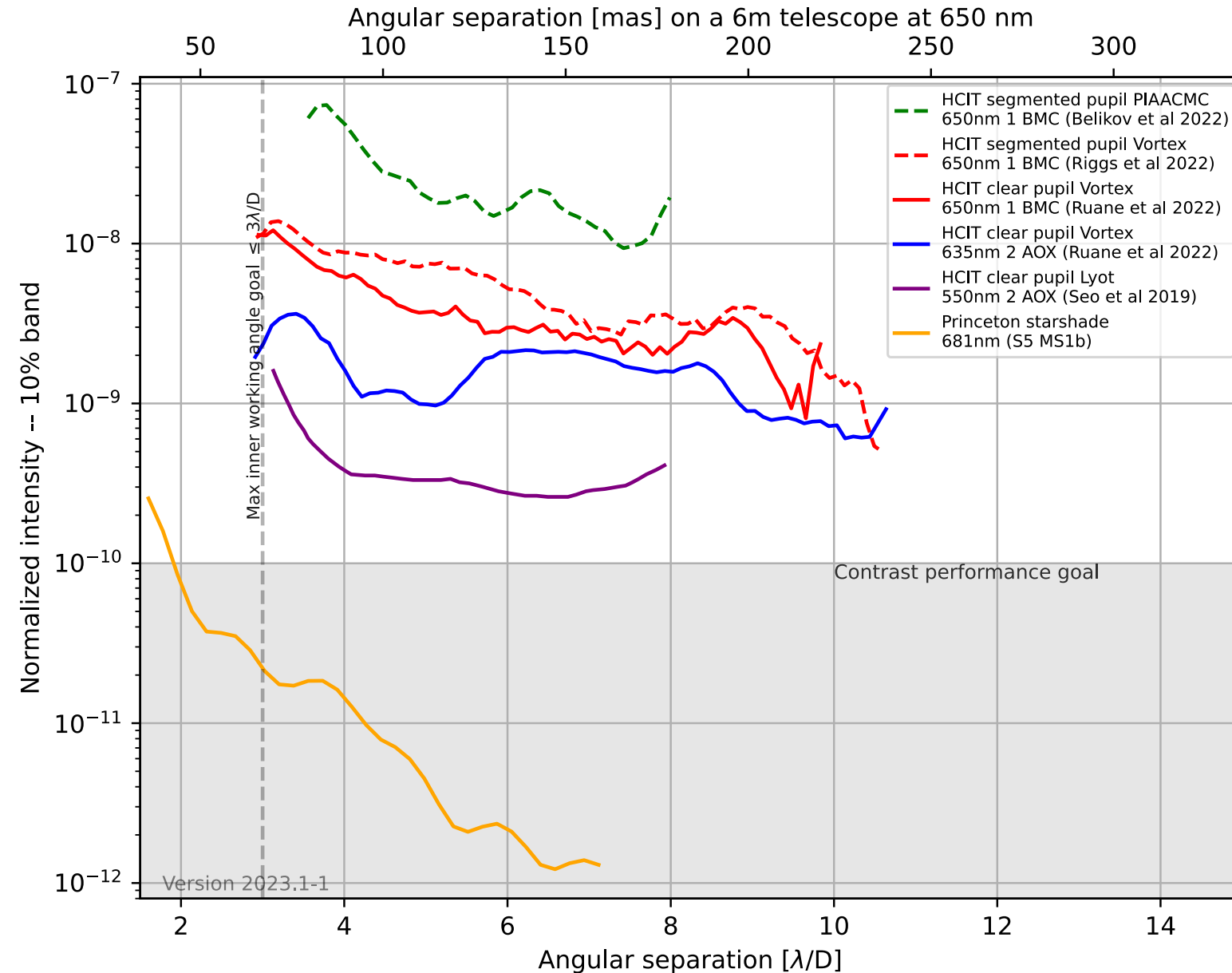
LMAPD NIR photon counting



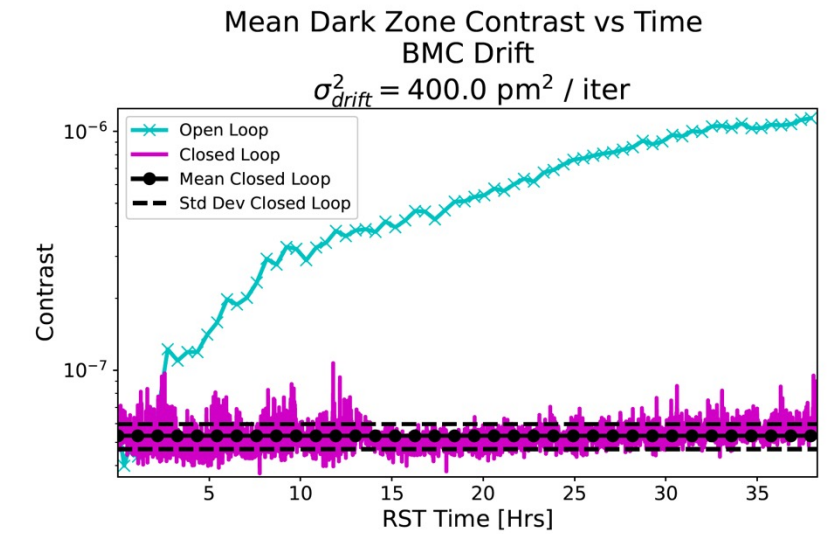
TRL 4

TRL 4 Microchannel Plate Detector for UVS

Starlight Suppression Performance



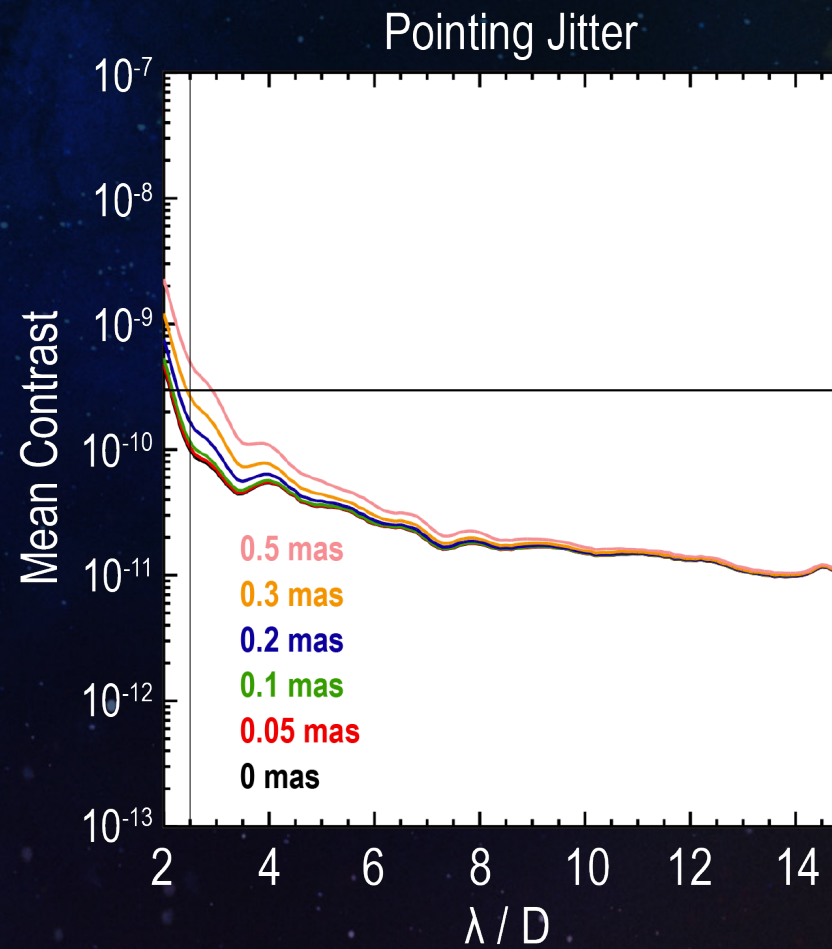
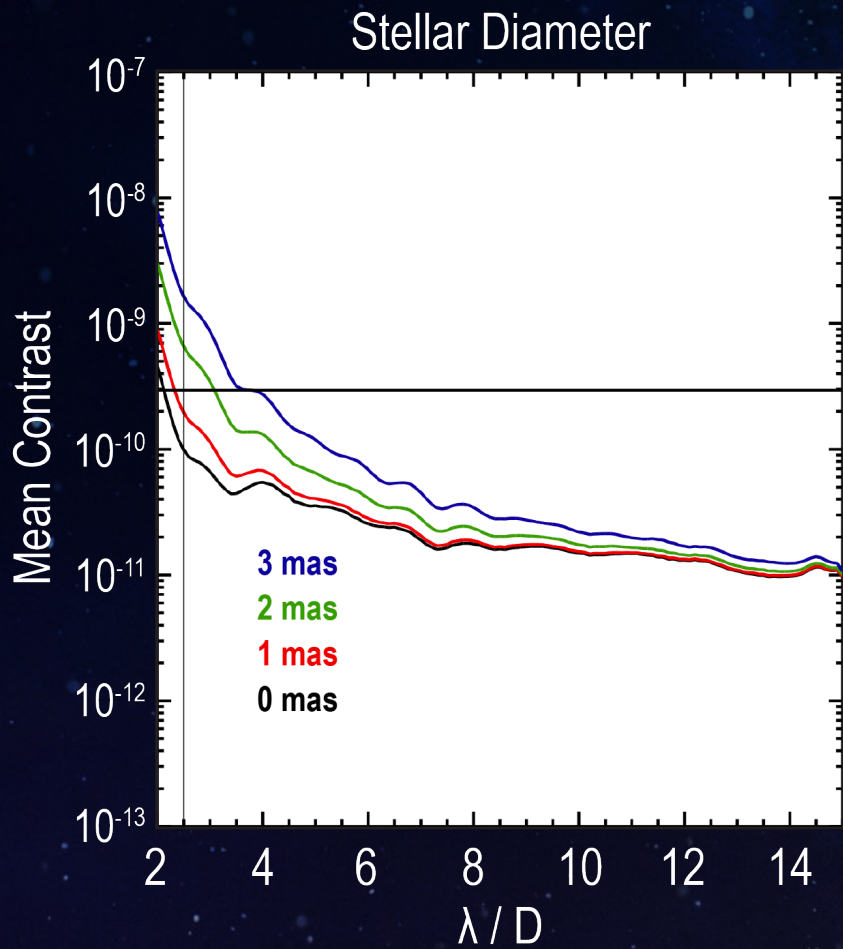
open-air HiCAT, segmented pupil PAPLC monochromatic, 1 BMC (Soummer et al 2022)



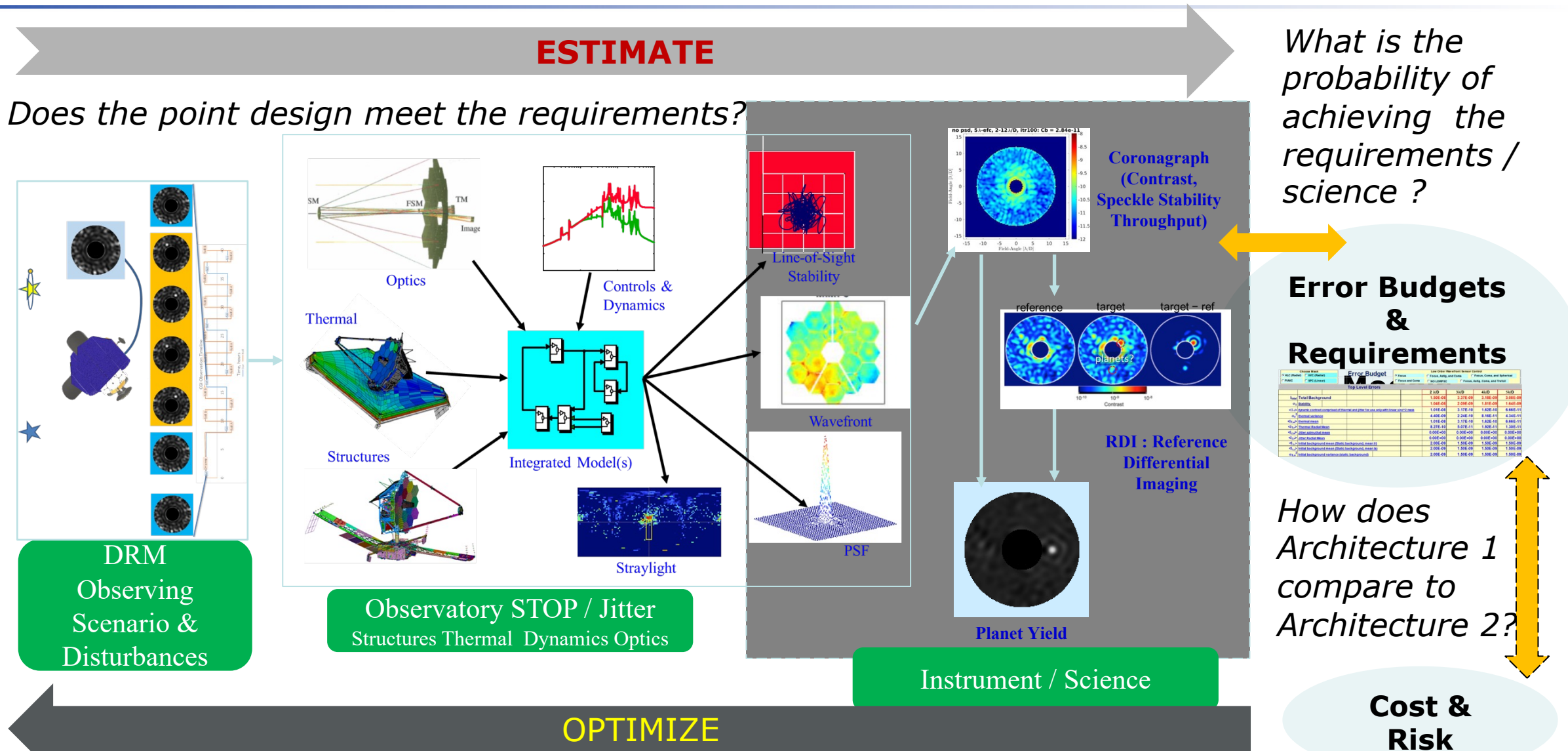
PAPLC active WFSC in dark hole (Redmond et al 2022)



HabEx Coronagraph Modeling Results



Modeling Threads to be Integrated for System Modeling



Metrics

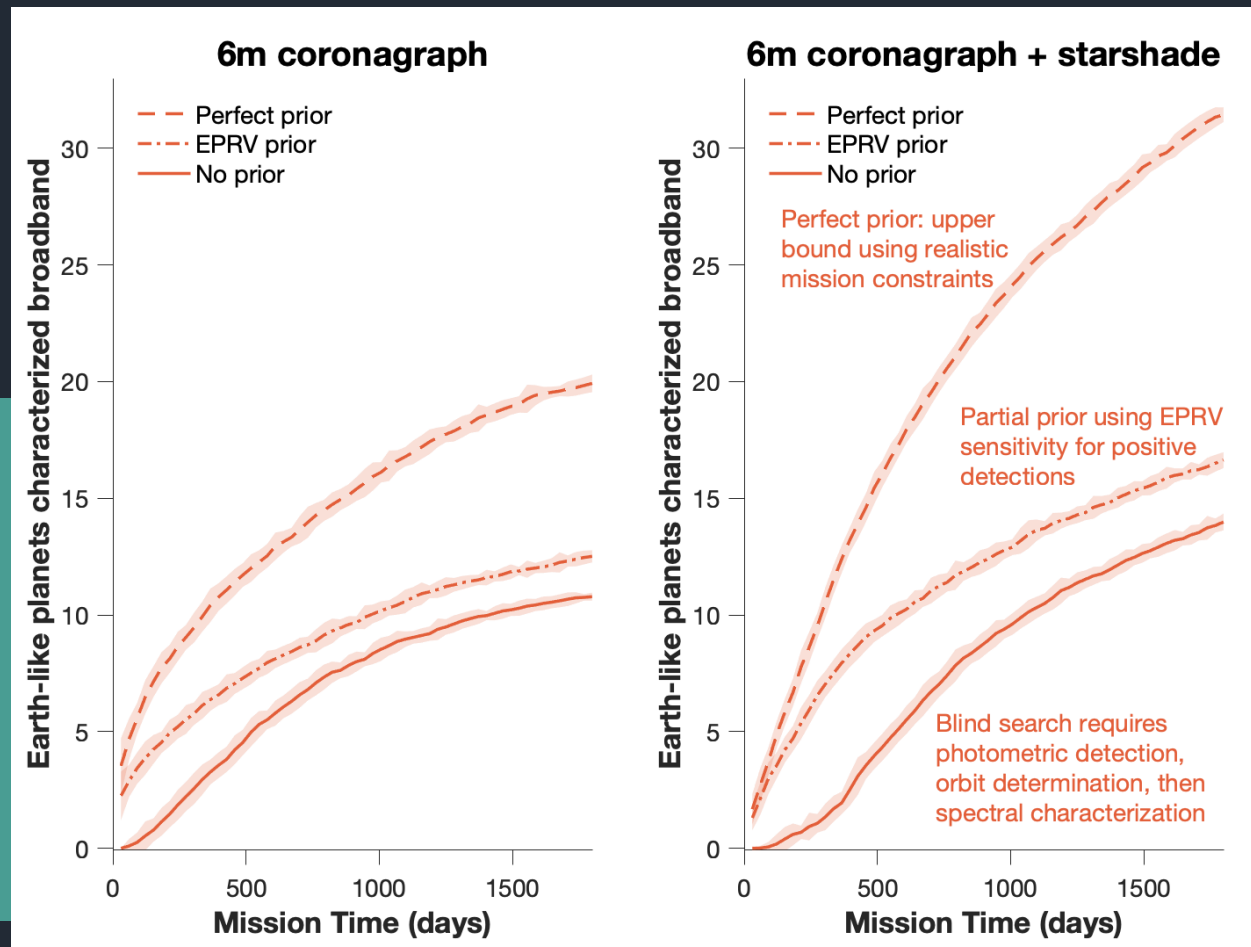
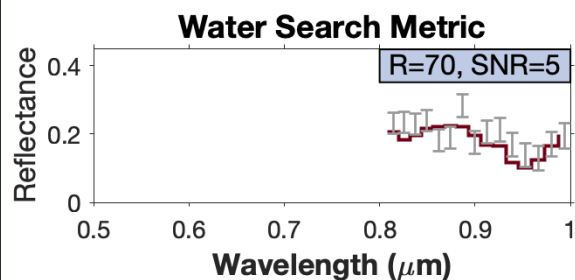
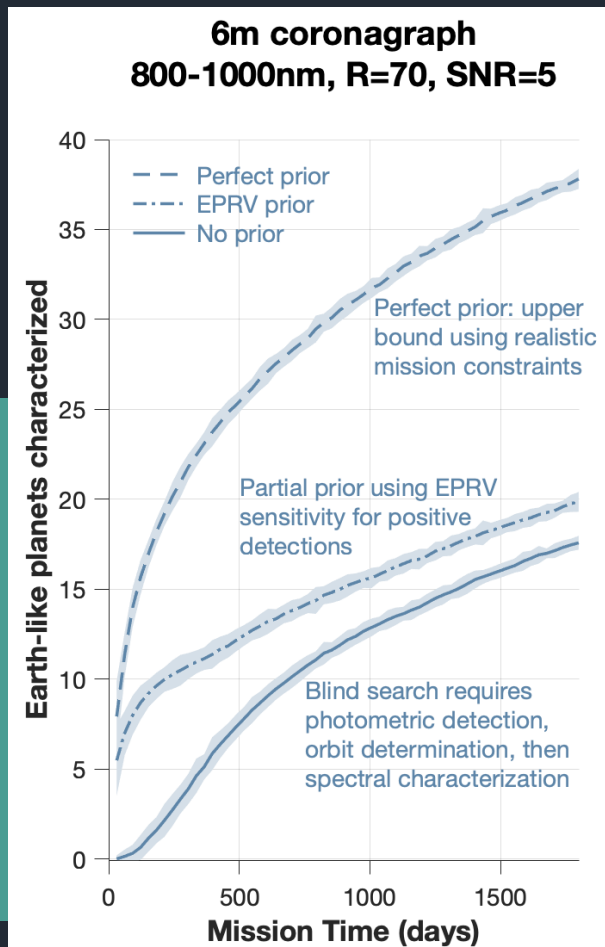
Bandwidth, SNR, R_s

Architecture

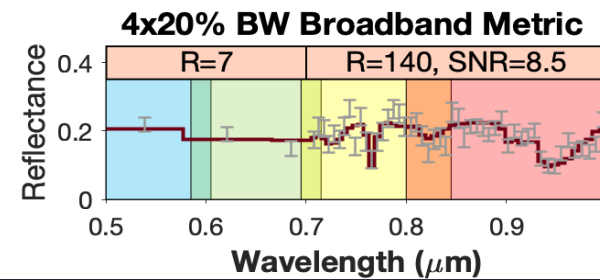
Observing Scenario

Prior Knowledge

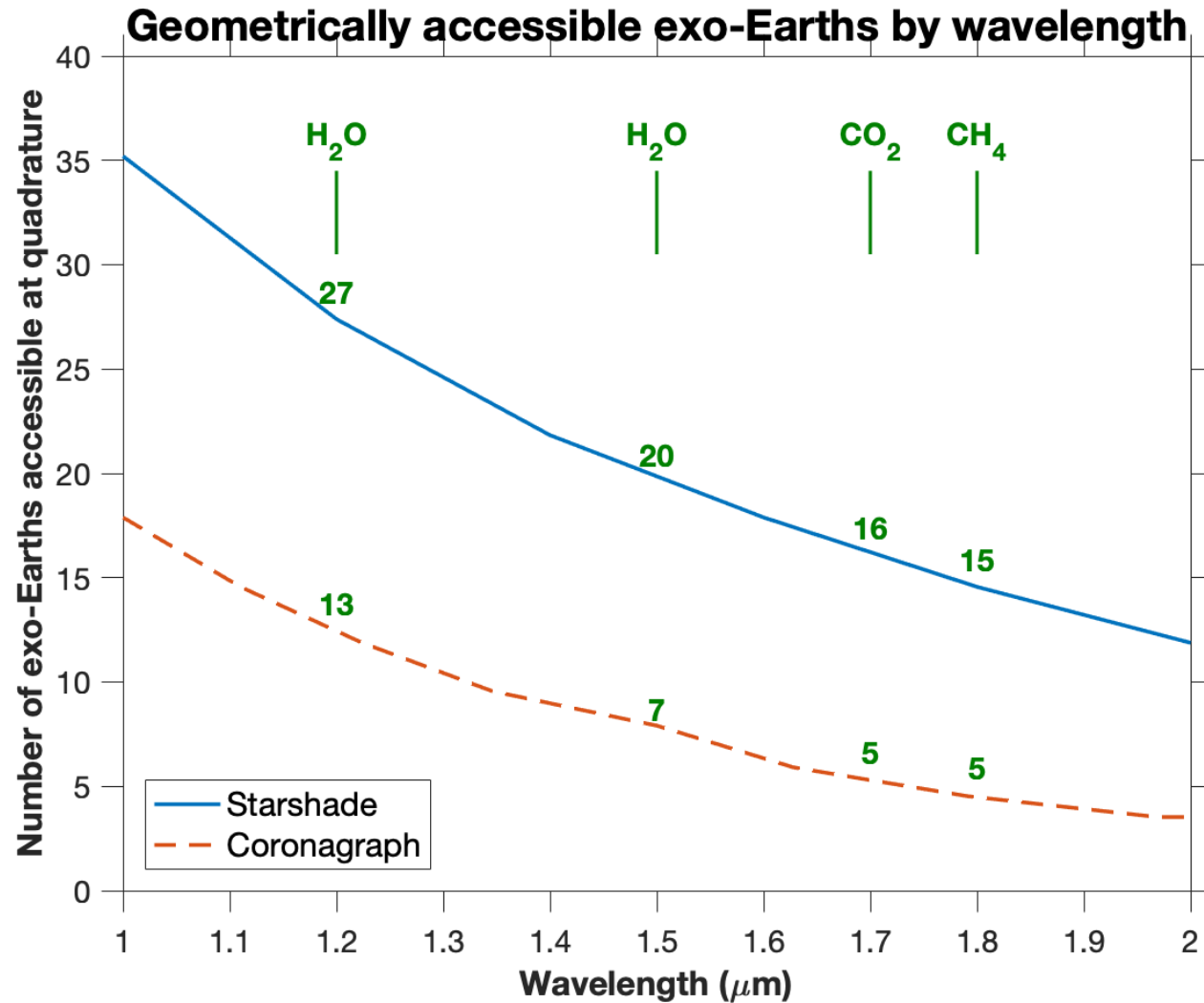
Different yield metrics reveal different sensitivities



Morgan et al 2022



Upper bound on yield at NIR wavelengths



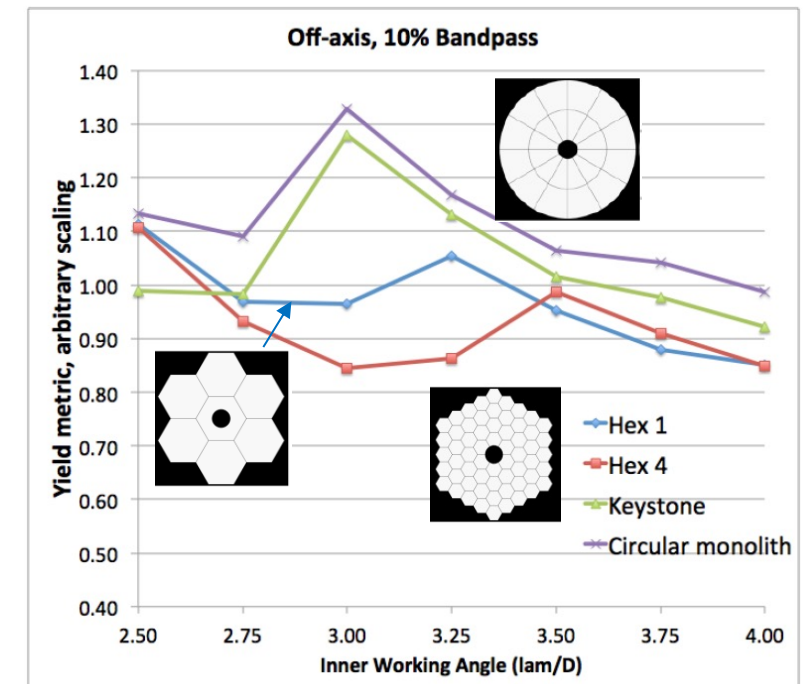
- Molecular species wavelength indicated where the red edge returns to the continuum
- Relative yield is due to IWA:
 - 1.56 I/D starshade
 - 2.5 I/D coronagraph

Morgan et al AAS 2023

Modeled using EXOSIMS <https://github.com/dsavransky/EXOSIMS>

Conclusions

- Coronagraphs require a system level solution
 - Coronagraph contrast AND throughput are important, as robustness to aberrations and dynamics
 - Inner working angle (IWA) strongly improves yield
 - Size of DM sets the outer working angle (OWA)
-
- **New possibilities**
 - Scalar Vortex coronagraphs for manufacturability
 - Photonic Lanterns (could they mature in time for HWO?)
 - Incorporate EPRV
 - Multi-star wavefront control
 - Segment shape for throughput and dynamics resilience
 - Detailed study of UV and NIR coronagraphs
 - Evaluate NIR and UV science requirements



https://exoplanets.nasa.gov/exep/technology/tech_colloquium/

BACKUP



UV Imager & Spectrograph (UVS)

Section 6.5

Imaging Channel	115 - 370 nm
Spectroscopy Channel	<ul style="list-style-type: none"> 115 - 320 nm with R=500 to 60,000 320 - 370 nm with R=500 or 1,000
Field of View	<ul style="list-style-type: none"> 3 x 3 arcmin² Micro-shutter Array for MOS: 2 x 2 array of 171 x 365 apertures
Effective Collecting Area	10x HST/COS

Baseline	Vector Vortex (Charge 6)
Visible Channels (1 per Polarization)	450 - 975 nm Imager + IFS with R=140
Near Infrared Channel	975 - 1800 nm Imager + IFS with R=40
High Contrast Region	IWA = 2.4 λ/D (62 mas at 0.5 μm) OWA = 32 λ/D (830 mas at 0.5 μm)
Raw Contrast	2.5 x 10 ⁻¹⁰ at IWA over 20% Bandwidth 40x better than WFIRST CGI
Features	Active Low Order Wavefront Sensing & Control with two 64x64 DMs

Coronagraph (HCG)

Section 6.3

Starshade Instrument (SSI)

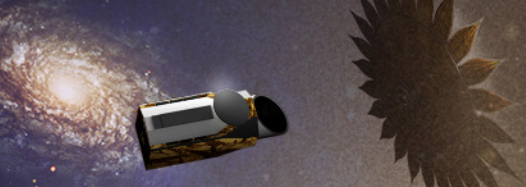
Section 6.4

UV Channel	200 to 450 nm Imager + Grism at R=7
Visible Channel	450 - 975 nm Imager + IFS with R=140
Near Infrared Channel	975 - 1800 nm Imager + IFS with R=40
High Contrast Region	IWA = 58 mas (from 300 to 1000 nm) OWA = 6" (Imager) / 1" (IFS)
Raw Contrast	10 ⁻¹⁰ at IWA over 107% Bandwidth (nominally 300 to 1000 nm)

Visible Channel	370 - 975 nm Imager + Grism with R=1000 >2x better resolution than HST < 600 nm
Near Infrared Channel	975 - 1800 nm Imager + Grism with R=1000
Field of View	<ul style="list-style-type: none"> 3 x 3 arcmin² Micro-shutter Array for MOS: 2 x 2 array of 171 x 365 apertures

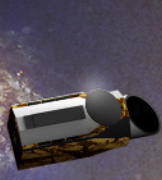
Workhorse Camera & Spectrograph (HWC)

Section 6.6



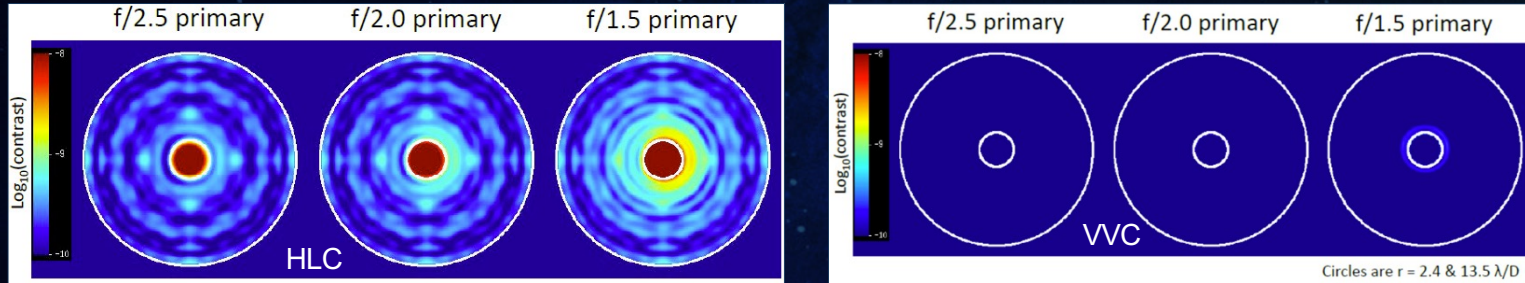
HabEx Science Goals & Objectives		HabEx Mission Architectures								
		4H	4C	4S	3.2H	3.2C	3.2S	2.4H	2.4C	2.4S
Habitable Exoplanets	O1	Green	Green	Green	Green	Green	Yellow	Yellow	Red	Red
	O2	Green	Green	Green	Green	Green	Yellow	Yellow	Red	Red
	O3	Green	Yellow	Green	Green	Green	Yellow	Yellow	Red	Red
	O4	Green	Yellow	Green	Green	Red	Yellow	Yellow	Red	Red
Exoplanetary Systems	O5	Green	Yellow	Red	Green	Yellow	Red	Yellow	Red	Red
	O6	Green	Yellow	Green	Green	Red	Yellow	Red	Red	Red
	O7	Green	Yellow	Green	Green	Yellow	Yellow	Red	Red	Red
	O8	Green	Yellow	Green	Green	Yellow	Green	Yellow	Yellow	Green
Observatory Science	O9	Green	Green	Green	Yellow	Yellow	Yellow	Yellow	Yellow	Yellow
	O10	Green	Green	Green	Yellow	Yellow	Yellow	Yellow	Yellow	Yellow
	O11	Green	Green	Green	Yellow	Yellow	Yellow	Yellow	Yellow	Yellow
	O12	Green	Green	Green	Green	Green	Green	Yellow	Yellow	Yellow
	O13	Green	Green	Green	Yellow	Yellow	Yellow	Red	Red	Red
	O14	Green	Green	Green	Green	Green	Green	Yellow	Yellow	Yellow
	O15	Green	Green	Green	Yellow	Yellow	Yellow	Red	Red	Red
	O16	Green	Green	Yellow	Green	Green	Green	Yellow	Yellow	Yellow
	O17	Green	Green	Green	Green	Green	Green	Green	Green	Green
Estimated Cost (\$B FY20)		6.8	4.8	5.7	5.7	3.7	5.0	4.8	3.1	4.0
Number of TRL4		13	10	9	12	9	9	11	8	8
Exo-Earths Characterized		8	5	5	5	3	4	3	1	2
Exoplanet Detections (all)		178	114	140	105	83	119	76	27	67

- *STDT's preferred architecture is 4H*
- *Red does not mean "no science"*
- *At a given size, Hybrid architectures maximize exoplanet science*
- *C-only*
 - *no UV exoplanet observations*
 - *Vast majority of planets with orbits*
 - *Reduced spectroscopy*
- *S-only:*
 - *High Quality spectra*
 - *Limited # of orbits measured*
- *Observatory Science is primarily a function of telescope size*
- *Architectures 4H (4C) and 3.2S studied in detail and "TRACEable"*



- Selecting among different coronagraph masks:

Figure 5.4-1



- Selecting from different VVC

Aberration	Indices		Allowable RMS wavefront error (nm) per mode			
	n	m	charge 4	charge 6	charge 8	charge 10
Tip-tilt	1	± 1	1.1	6.1	16	29
Defocus	2	0	0.8	4.6	13	32
Astigmatism	2	± 2	0.0068	1.1	0.92	4.8
Coma	3	± 1	0.0064	0.69	0.84	5.4
Spherical	4	0	0.0049	0.53	0.75	7
Trefoil	3	± 3	0.0073	0.0064	0.59	0.68
Exo-Earths Spectra			9	8	5	3

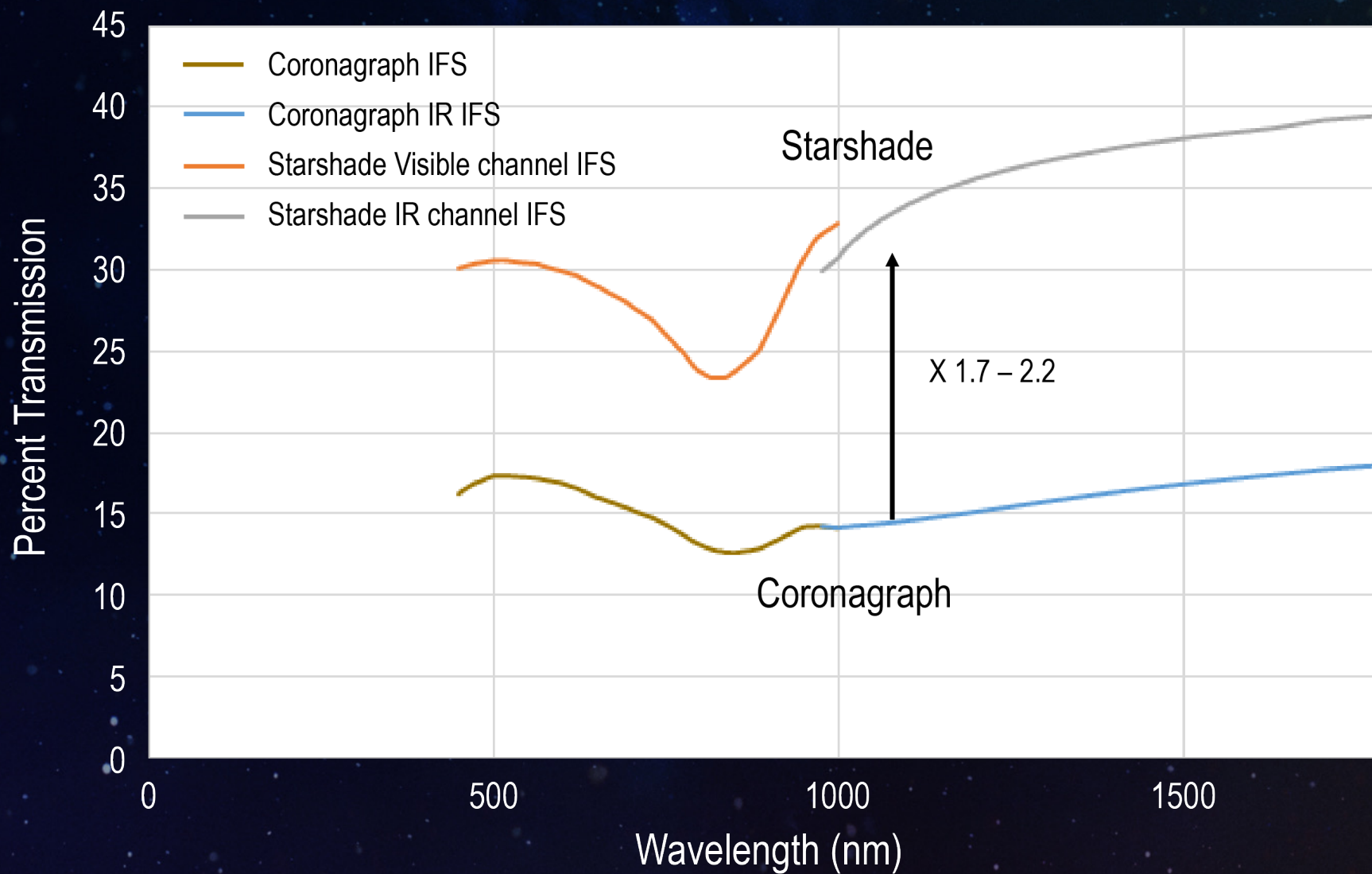
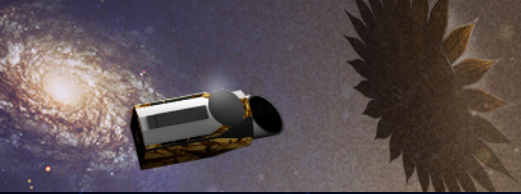


Aperture Diameter	4.0 meters
Diffraction Limited Wavelength	400 nm
Total Wavefront Error	30 nm rms (wavefront)
Total Primary Mirror Figure Error	5.6 nm rms (surface)
Low Spatial SFE (<30 cycles/dia)	4.3 nm rms
Mid Spatial SFE (30 to 100 cycles/dia)	3.3 nm rms
High Spatial SFE (>100 cycles/dia)	1.4 nm rms
Roughness	0.3 nm rms
Line of Sight Stability (Jitter)	< 0.7 milli-arcseconds
Wavefront Error Stability	1 to 250 pm (spatial frequency dependent)
Spectral Range	115 nm to 1700 nm

Specification	Predicted Margin	Enabling Design Elements
LOS Mechanical Jitter	20X	Telescope Structure Stiffness Low-Noise Micro-Thrusters
LOS Thermal Drift	3.5X	Laser Metrology System
Diffraction-Limited Transmitted Wavefront	1X	Demonstrated Mirror Fabrication Capability Laser Metrology System for Alignment
Wavefront Stability	4X	Telescope Structure Stiffness PM Substrate Stiffness, CTE and Thermal Mass Active Thermal Control, Low-Noise Micro-Thrusters



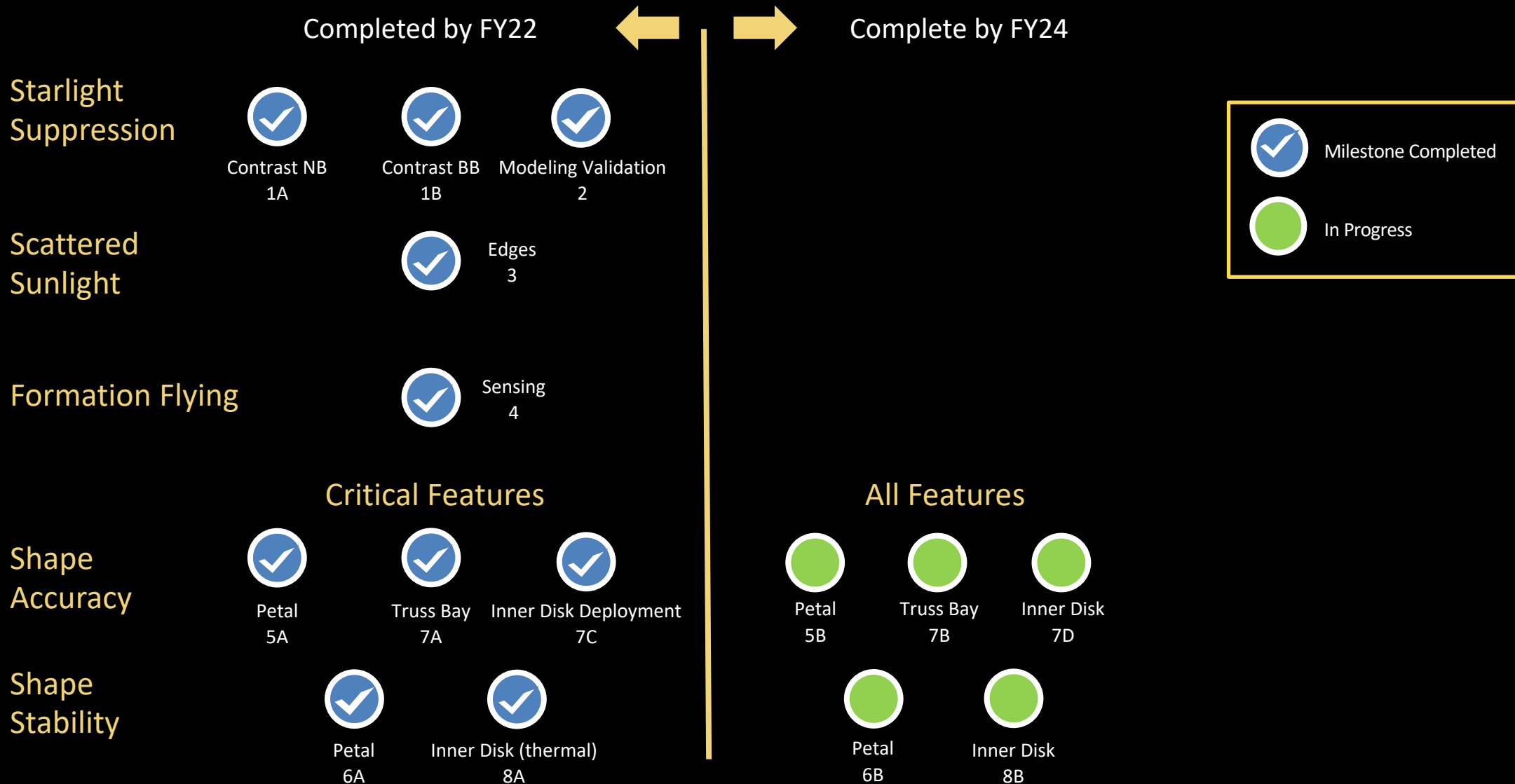
Parameter	Requirement	Expected Performance	Margin
Observational band	0.30–1.7 μm	0.20–1.80 μm	Met by design
IWA	≤ 64 mas (0.87 μm)	57 mas (0.87 μm)	12% (0.87 μm)
	≤ 80 mas (1.0 μm)	58 mas (1.0 μm)	38% (1.0 μm)
Raw contrast	$\leq 1.0 \times 10^{-10}$	6.0×10^{-11}	67%
Raw contrast stability	$\leq 2.0 \times 10^{-11}$	1.0×10^{-11}	100%
Pointing control	$\leq 1^\circ$	$\ll 1^\circ$	Met by design
Solar edge scatter	$V > 25$ mag/arcsec ²	$V > 25$ mag/arcsec ²	Met by design
Sunlight leakage	$> 32 V_{\text{mag}}$	$> 32 V_{\text{mag}}$	Met by design
Micrometeoroid holes	≤ 500 ppm	5 ppm	9900%
Petal position (manufacture)	$\leq \pm 600$ μm	± 340 μm	76%
Petal shape (manufacture)	$\leq \pm 140$ μm	± 80 μm	75%
Petal position (thermal)	$\leq \pm 400$ μm	± 62 μm	545%
Petal shape (thermal)	$\leq \pm 160$ μm	± 50 μm	220%





S5: Closing Starshade Technology Gaps

<https://exoplanets.nasa.gov/exep/technology/starshade/>



What is Integrated Modeling (IM) ?

- IM refers to the pipeline of engineering model and analyses required to verify observatory performance based on Systems Error Budget (EB) Metrics
- IM provides inputs as estimates into Systems Error Budget allocations and Science yield for Architecture Trades
- Typical IM Disciplines in the pipeline:
 - Thermal, structures, dynamics, attitude control, optics with wavefront sensing and control, straylight, coronagraph models
- Typical Error Budget Metrics computed with IM
 - WFE, WFE stability (thermal drift & jitter), Pointing stability, Alignments, PSF, Contrast
- Note: Science Yield models could also be integrated into the IM pipeline